

Fig. 2 Variation of SID with HID.

Thus, Eq. (3) is approximately four times more accurate than Eq. (1), which is twice less accurate than Eq. (2). The superiority of the correlation given by Eq. (3) over the other correlations is more apparent for small values of H/δ and H/D: $\sigma_1 = 0.70$ and $\sigma_2 = 0.34$ for $H/\delta < 2$, whereas $\sigma_3 = 0.13$ for H/D < 1.

Two important conclusions must be restated: 1) the cylinder diameter D is the appropriate scaling parameter for this flow and 2) the correlations given by Eqs. (1) and (2) are not as accurate as the correlation of Sedney and Kitchens.²

References

¹Verma, S. B., and Gupta, V., "Supersonic Separation with Obstructions," *AIAA Journal*, Vol. 34, No. 4, 1996, pp. 849–851.

²Sedney, R., and Kitchens, C. W., "Separation Ahead of Protuberances

²Sedney, R., and Kitchens, C. W., "Separation Ahead of Protuberances in Supersonic Turbulent Boundary Layers," *AIAA Journal*, Vol. 15, No. 4, 1977, pp. 546–552.

³Westkaemper, J. C., "Turbulent Boundary-Layer Separation Ahead of Cylinders," *AIAA Journal*, Vol. 6, No. 7, 1968, pp. 1352–1355.

⁴Settles, G. S., and Dolling, D. S., "Swept Shock-Wave/Boundary-Layer Interactions," *Tactical Missile Aerodynamics*, edited by M. J. Hemsch and J. N. Nielsen, Vol. 104, Progress in Astronautics and Aeronautics, AIAA, New York, 1986, pp. 297–379.

⁵Dolling, D. S., and Rodi, P. E., "Upstream Influence and Separation Scales in Fin-Induced Shock Turbulent Boundary-Layer Interaction," *Journal of Spacecraft and Rockets*, Vol. 25, No. 2, 1988, pp. 102–108.

⁶Hung, C., and Buning, P. G., "Simulation of Blunt-Fin Induced Shock-Wave and Turbulent Boundary-Layer Interaction," *Journal of Fluid Mechanics*, Vol. 154, May 1985, pp. 163–185.

⁷Özcan, O., and Yüceil, B.K., "Cylinder-Induced Shock-Wave Boundary-Layer Interaction," *AIAA Journal*, Vol. 30, No. 4, 1992, pp. 1130–1132.

⁸Özcan, O., "An Experimental Investigation of Three-Dimensional Boundary-Layer Separation in Supersonic Flow Past a Circular Cylinder on a Flat Plate," Ph.D. Dissertation, Dept. of Mechanical Engineering, Univ. of California, Berkeley, CA, March 1982.

⁹Yüceil, K. B., "An Experimental Investigation of Shock-Wave/Boundary-Layer Interaction Induced by Cylindrical Obstacles," M.S. Thesis, Dept. of Aeronautics and Astronautics, Istanbul Technical Univ., Istanbul, Turkey, June 1989

P. R. Bandyopadhyay Associate Editor

Reply by the Authors to O. Özcan and K. B. Yüceil

S. B. Verma*and Vijay Gupta[†] *Indian Institute of Technology, Kanpur 208 016, India*

The authors would like to thank Özcan and Yüceil for their comments. The exponential curve fit of Eq. (2) in their comments is quite clearly a better equation. The authors proposed Eq. (1) because it was quite adequate for the data available to them. However, the contention of Ref. 1 that the cylinder diameter D is the appropriate scaling parameter for this flow is not borne out by the evidence presented by them. The true measure of the relative scatter is not the standard deviation σ in cases where different variables are being considered, as in the present case, but σ normalized with a characteristic value of the relevant variables. The values of δD are fairly small so that the values of the variables in Fig. 1 of Ref. 1 are much larger than the corresponding values in their Fig. 2. Then again, most of the new points^{2,3} (from their Refs. 8 and 9), which appear as quite scattered on Fig. 1, are lying mostly below the curve for Eq. (3) in Fig. 2 (see Ref. 1).

The discussor's evidence is, therefore, not conclusive. Further, from a physical point of view alone, the characteristic length for separation distance should be derivable from blockage area rather than *D*.

References

¹Özcan, O., and Yüceil, K. B., "Comment on Supersonic Separation with Obstructions," *AIAA Journal*, Vol. 35, No. 8, 1997, pp. 1423, 1424.

²Özcan, O., "An Experimental Investigation of Three-Dimensional Boundary-Layer Separation in Supersonic Flow Past a Circular Cylinder on a Flat Plate," Ph.D. Dissertation, Dept. of Mechanical Engineering, Univ. of California Berkeley, CA. March 1982

of California, Berkeley, CA, March 1982.

³Yüceil, K. B., "An Experimental Investigation of Shock-Wave/Boundary-Layer Interaction Induced by Cylindrical Obstacles," M.S. Thesis, Dept. of Aeronautics and Astronautics, Istanbul Technical Univ., Istanbul, Turkey, June 1989.

P. R. Bandyopadhyay Associate Editor

Received March 10, 1997; accepted for publication March 10, 1997. Copyright 997 by the American Institute of Aeronautics and Astronautics, Inc. All rights reserved.

^{*}Research Scholar, Department of Aerospace Engineering.

[†]Professor, Department of Aerospace Engineering.